			Mi	lestone	e Reviev	v Flyshe	et				
Institution	titution Northe			eastern University			Milestone		Critial Design Revie		
Vehicle Properties						Motor Properties					
Total Leng	gth (in)		140			Motor Designation Cesard			oni L1395 Blue	estreak	
Diamete	er (in)		6.16			Max/Averag	e Thrust (lb)		404.66/328.9	)	
Gross Lift Off	Weigh (lb)		45.713			Total Impu	ulse (lbf-s)		1101.46		
Airframe N	/laterial		BlueTube			Mass Before	e/After Burn	4.323 kg b	efore and 1.8	48 kg after	
Fin Mat	erial	(	G10 Fiberglass			Liftoff Th	nrust (lb)		359.694		
Coupler Length 2 x 12 inc   Coupler Length overlap, 2		ich coupler with 6 inch x 14 inch coupler with 7 inch overap			Motor Retention Aeropac		Aeropack	k 75mm Motor Retainer			
	Sta	bility Anal	ysis				As	cent Analy	sis		
Center of Pr	ressure (in fi	om nose)	101.23 in.			Maximum Veloxity (ft/s)		(ft/s)	650		
Center of G	Gravity (in fr	om nose)	85.01 in.			Maximum Mach Number			0.58		
Static Stability Margin			5.224 calibers of stability			Maximum Acceleration (ft/s^2)		n (ft/s^2)		248.15	
Static Stability	v Margin (of	launch rail)	4.258 calibers of stability			Target Apogee (From Simulations)			5318 ft		
Thrust	Thrust-to-Weight Ratio			7.125		Stable Velocity (ft/s)			52		
Rail Siz	e and Lengt	h (in)	1.5 X 1.5 in rail with length of 144 in			Distance to Stable Velocity (ft)		ocity (ft)	6.25		
Rai	l Exit Velocit	Ξ¥	73.82 feet p	er second							
	Recovery	System P	roperties			Recovery System Properties					
	Dro	gue Parach	nute			Main Parachute					
Manufacture	er/Model	Fruity Ch	y Chutes Compact Elliptical Parachute			Manufacturer/Model		Fruity Chutes Iris Ultra Light Parachute			
Size	2		18 in.			Size		Payload Section - 60 in. Booster Stage - 66 in.			
Altitude at Deployment (ft)			Payload Section - 5280 ft Booster Stage - 5173 ft			Altitude at Deployment (ft)		ent (ft)	Payload Section - 300 ft Booster Stage - 500 ft		
Velocity at Deployment (ft/s)			Payload Section - 0 ft per sec Booster Stage - 25 ft per sec			Velocity at Deployment (ft/s)		nt (ft/s)	Payload Section - 80 ft per sec Booster Stage - 90 ft per sec		
Terminal Velocity (ft/s)			Payload Section - 80 ft per sec Booster Stage - 90 ft per sec			Terminal Velocity (ft/s)		Payload Section - 18 ft per sec Booster Stage - 19 ft per sec			

Recove	Recovery Harness Material		Kevlar		Rec	Recovery Harness Material			Kevlar	
Harnes	Harness Size/Thickness (in)			1/4		Harness Size/Thickness (in)		1/2		
Recover	Recovery Harness Length (ft)			15		Recovery Harness Length (ft)		40		
Harness/ Inter	Harness/Airframe Interfaces 1		l/4 in - 20 eyebolt		Harne	Harness/Airframe Interfaces		1/2 - 13 hoist ring with 2 in. washers		
Kinetic Energy of	Section 1	Section 2	Section 3	Section 4	Kinetic Energy (	Section 1	Section 2	Section 3	Section 4	
Each Section (Ft- Ibs)	25.99	61.68	71.74	65.11	Each Section ( Ibs)	Ft- 25.99	61.68	71.74	65.11	

Reco	very Electronics	Reco	very Electronics	
Altimeter(s)/Timer(s) (Make/Model)	Perfect Flite StratoLogger CF	Rocket Locators (Make/Model)	XBEE Pro XSC (S3), BigRedBee 2 M High Power GPS	
Dedundancy Dian		Transmitting Frequencies	900 Mhz, 900 Mhz	
Redundancy Plan	2 StratoLoggers per black powder charge	Black Powder Mass Drogue Chute (grams)	Payload Section - 2 x 1.3 Booster Stage - 2 x 1.3	
Pad Stay Time (Launch Configuration)	according to StratoLogger manual	Black Powder Mass Main Chute (grams)	Payload Section - 2 x 1.3 Booster Stage - 2 x 1.3	

## Milestone Review Flysheet

Milestone

Institution

Autonomous Ground Support Equipment (MAV Teams Only)											
	Overview										
Capture Mechanis m											
	Overview										
Container Mechanis m											
					Over	view					

Launch Rail Mechanis m	***Include Description of rail locking mechanism***												
					Over	view							
Igniter Installation Mechanis m													
					Payload								
					Over	view							
Payload 1	Our payload is a very passive system because we wanted to minimize failure of the system. The payload should not leave the launch vehicle at any point during the launch. It will be recovered with the rocket via parachute and should not itself affect safety of the launch vehicle												
					Over	view							
Payload 2													
				Test Plans	s, Status, ai	nd Results							
Ejection Charge Tests				Will sta	tically ejectio	n test all para	ichutes.						
Sub-scale Test Flights													
			3	Sub-scale lest	to prove aero	Juynannic Stat	Jiilty of design	1					
Full-scale Test Flights													
		Full	-scale to laun	ch in Febuary	/March to pro	ove stability a	nd structual i	ntegrity of de	esign	1			
			M	ilestone	e Reviev	v Flyshe	et						

Institution	Milestone
Additional Com	ments
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